J. Kliber; I. Schindler Desctiption of stress-strain curves on high temperture deformed steels

Acta Universitatis Carolinae. Mathematica et Physica, Vol. 32 (1991), No. 1, 95--101

Persistent URL: http://dml.cz/dmlcz/142633

Terms of use:

© Univerzita Karlova v Praze, 1991

Institute of Mathematics of the Academy of Sciences of the Czech Republic provides access to digitized documents strictly for personal use. Each copy of any part of this document must contain these *Terms of use*.



This paper has been digitized, optimized for electronic delivery and stamped with digital signature within the project *DML-CZ: The Czech Digital Mathematics Library* http://project.dml.cz

Description of Stress-Strain Curves on High Temperature Deformed Steels

J. KLIBER, I. SCHINDLER*)

Czechoslovakia

Received 27 August 1990

Introduction

In response to the growing interest on the work hardening and softening processes of the metallic material under the heat loading in the zone of plastic deformation, a theoretical and laboratory study has been conducted to examine and to define in mathematical form the shape of stress-strain curve.

The stress-strain curve of a polycrystalline metal [1, 2] under any simple loading has the shape shown diagrammatically in current pictures. The three most usual types of test are the tension of a rod, the compression of a short cylindrical block, and the twisting of a thin-walled tube or a prismatic bar. The result of such a test is represented by plotting the mean stress σ (tensile, compressive or shear) acting over the current (initial) cross-sectional area, against some measure of the total strain. The amout of deformation is usually measured as equivalent (logharitmic = = true) strain.

Over the years, a number of investigations defined the overal response of a material in terms fo some generalized equation of state

$$\sigma = f(T, \varepsilon, \dot{\varepsilon}) \tag{1}$$

e.g. on the basis of relaxation's study proposed [3]

$$\sigma = \sigma_0 \cdot B_1 \exp(n_1 T) \cdot B_2 \varepsilon^{n_2} \cdot \exp(n_3 \varepsilon) .$$
 (2)

 $B_3\dot{\varepsilon}^{n_4}$. $B_4\tau^{n_5}$. exp $(n_6\tau)$

where B, to B_4 nad n_1 to n_6 are constants, T, ε , $\dot{\varepsilon}$, τ are temperature (°C or K), strain, strain rate (s⁻¹) and dwell period between deformations in (s), respectively. Other possibility is [4]

$$\sigma = \sigma_0 \cdot \varepsilon^n \cdot \dot{\varepsilon}^m \cdot \exp\left(C + A/T\right). \tag{3}$$

At present the description resulted in [5]

$$\sigma = A \cdot \varepsilon^{b} \cdot \exp\left(-c\varepsilon^{d}\right) \cdot \dot{\varepsilon}^{fT} \cdot \exp\left(g/T\right).$$
(4)

^{*)} Faculty of Metallurgy and Material Eng., Ostrava, Czechoslovakia.

Stress-strain curves

PEAK STRESS

Characteristic stresses (original σ_{ss} – stress at sready-state condition) may be correlated with temperature and strain rate. At low stresses ($\alpha\sigma_{ss} < 0.8$) and high stresses ($\alpha\sigma_{ss} < 1.2$) lead these equations in common form [6]

$$\dot{\varepsilon}_{ss} = A \exp\left(-Q/(RT)\right) \left[\sinh\left(\alpha\sigma_{ss}\right)\right]^{n}$$
(5)

where A, α , n are constants and Q is activation energy in (J/mol), and all data may be correlated using the Zener-Hollomon parameter $Z = \dot{\epsilon} \exp(Q/(RT))$. Relationship (5) has been applied successfully with σ_p – stress at peak (STRESSP) and transformed into [7, 8]

$$\sin h (\alpha \sigma_p) = (Z/A)^{1/n}$$
(6)

$$\sigma_p = \frac{1}{\alpha} \arg \sin h \left(\frac{\dot{\varepsilon} \cdot \exp\left(Q/(RT)\right)}{A} \right)^{1/n}$$
(7)

PEAK STRAIN

Temperature compensated strain rate Z is related to time compensated strain rate W as [9]

$$W = A_p \cdot Z^{-a_p} . \tag{8}$$

Hence, using $\varepsilon_p = W_p \cdot Z$ [10] and eq. (8)

$$\ln \varepsilon_p = \ln A_3 + (1 - a_3) \ln \dot{\varepsilon} + (1 - a_3) \frac{Q}{RT}$$
(9)

is obtained where A_p , A_3 , a_3 are constants, $W_p = t_p \exp(-Q/(RT))$ and t_p is time to peak in (s). By gradual linear regression analysis we become the slope $(1 - a_3)$. For expressing the thermally and strain rate activated process on peak strain ε_p a newly suggested and verified form have been obtained as follows [10, 11]

$$\varepsilon_p = \dot{\varepsilon}^a (P + \sqrt{(P^2 + 1)}) \tag{10}$$

where P is temperature dependent, P = Y + X/T, Y, X are constants, a has the same meaning as $(1 - a_3)$ eq. (9) and with rearrangement is obtained

$$\varepsilon_p = \dot{\varepsilon}^a \cdot \exp \arg \sin h \left(Y + X/T \right).$$
 (11)

STRESS

For a fixed T, the stress-strain curves are progressively higher at medium $\varepsilon/\varepsilon_p$ for a higher strain rate. Although these curves are lower for decreasing Z value. The function which describes best the reduced flow stress is [12]

$$\ln\left(\frac{\sigma}{\sigma_p}\right) = c \left[\ln\left(\frac{\varepsilon}{\varepsilon_p}\right) + \left(1 - \frac{\varepsilon}{\varepsilon_p}\right)\right]$$
(12)

where c is a positive constant between zero and unity, which decreases as the curves are higher. The value of c is determined through plots of $\left[-\ln (\sigma/\sigma_p)\right]$ vs. $\left[-\ln (\varepsilon/\varepsilon_p) - (1 - \varepsilon/\varepsilon_p)\right]$. These should be linear but are never perfectly so.

The evaluation procedure

The result of continuous torsion tests are taken into the calculation – as torque moment and strain at various temperatures $(T \in (800; 1000)^{\circ}C \text{ and strain rates } \sigma \in (6; 1000) \text{ min}^{-1})$ from hot deformed HSLA steel. The specimens were 50 mm long having a diameter of 6 mm.

1. STRESSPC (STRESS at Peak Calculated)

The data can be fitted by

(13)
$$\sigma = 90 \arg \sin h \left(\frac{\dot{\varepsilon} \exp \frac{365 \, 400}{RT}}{8 \, . \, 10^{13}} \right)^{0.2}$$

using personal computer AT.

2. STRAINPC (STRAIN at Peak Calculated)

(14)
$$\varepsilon_{PC} = \dot{\varepsilon}^{0,142}$$
. exp arg sin h $\left(-6.9 + \frac{7\,180}{T}\right)$

3. STRESS (STRESS Experimental compared with STRESS Calculated)

(15)
$$\sigma_c = \sigma_{PC} \left[\frac{\varepsilon}{\varepsilon_{PC}} \exp\left(1 - \frac{\varepsilon}{\varepsilon_{PC}}\right) \right]^c.$$

Discussion

The value of exponent c, wich is variable, slightly complicate common use of equation (15). Taking into account that the resulting c is adequate for temperature,



Fig. 1. Effect of strain rate from 6-1000 min⁻¹ and true strain from 0.1 to 0.5 on STRESS Calculated (eq. 17).



Fig. 2. Effect of strain rate from $6-1000 \text{ min}^{-1}$ and true strain from 0.1 to 0.5 on STRESS Experimental.



Fig. 3. Comparison of STRESSE and STRESSC for $a \dots T = 1073$ K, various strain and strain rates $b \dots T = 1273$ K.

TEMPERATURE T = 1123 / K /RATE o = 6 / min - 1 / leps= .1 STRESS (EXPER.) = 68 /MPa/ (CALCUL.) = 78 /MPa/ eps= .2 STRESS (EXPER.) = 96 /MPa/ (CALCUL.) = 100 /MPa/ STRESS (EXPER.) = 105 /MPa/ eps= .3 (CALCUL.) = 107 /MPa/ eps= .4 STRESS (EXPER.) = 104 /MPa/ (CALCUL.) = 106 /MPa/ STRESS (EXPER.) = 98 /MPa/ eps= .5 (CALCUL.) = 100 /MPa/ /MPa/ (CALCUL.) = 93 /MPa/ eps= .6 STRESS (EXPER.) = 94(CALCUL.) = 84 /MPa/ eps= .7 STRESS (EXPER.) = 89/MPa/ STRESS (EXPER.) = 88/MPa/ (CALCUL.) = 76 /MPaú eps=.8TEMPERATURE T = 1123 / K /RATE o = 50 / min - 1/STRESS (EXPER.) = 78 /MPa/ eps= .1 (CALCUL.) = 86 /MPa/ STRESS (EXPER.) = 105 /MPa/ (CALCUL.) = 117 /MPa/ eps= .2 eps= .3 (CALCUL.) = 133 /MPa/ STRESS (EXPER.) = 128 /MPa/ STRESS (EXPER.) = 137 /MPa/ (CALCUL.) = 139 /MPa/ eps= .4 (CALCUL.) = 140 /MPa/ eps= .5 STRESS (EXPER.) = 139 /MPa/ eps= .6 (CALCUL.) = 137 /MPa/ STRESS (EXPER.) = 135 /MPa/ STRESS (EXPER.) = 130 /MPa/ (CALCUL.) = 132 /MPa/ eps= .7 STRESS (EXPER.) = 124 /MPa/ (CALCUL.) = 125 /MPa/ eps= .8 TEMPERATURE T = 1123 / K /RATE o = 500 /min-1/ (CALCUL.) = 90 /MPa/ (CALCUL.) = 127 /MPa/ eps= .1 STRESS (EXPER.) = 80 /MPa/ STRESS (EXPER.) = 120 /MPa/ eps= .2 STRESS (EXPER.) = 138 /MPa/ (CALCUL.) = 148 /MPa/ eps= .3 eps= .4 STRESS (EXPER.) = 155 /MPa/ (CALCUL.) = 161 /MPa/ STRESS (EXPER.) = 165 /MPa/ (CALCUL.) = 168 /MPa/ eps= .5 eps= .6 STRESS (EXPER.) = 168 /MPa/ (CALCUL.) = 170 /MPa/ eps= .7 STRESS (EXPER.) = 168 /MPa/ (CALCUL.) = 169 /MPa/ eps= .8 STRESS (EXPER.) = 160 /MPa/ (CALCUL.) = 166 /MPa/ TEMPERATURE T = 1123 / K /RATE o = 1000 / min - 1/(CALCUL.) = 93 /MPa/ (CALCUL.) = 133 /MPa/ STRESS (EXPER.) = 100 /MPa/ eps= .1 eps= .2 STRESS (EXPER.) = 142 /MPa/ eps= .3 STRESS (EXPER.) = 162 /MPa/ (CALCUL.) = 158 /MPa/ (CALCUL.) = 173 /MPa/ eps= .4 STRESS (EXPER.) = 175 /MPa/ eps= .5 STRESS (EXPER.) = 182 /MPa/ (CALCUL.) = 183 /MPa/ eps= .6 STRESS (EXPER.) = 186 /MPa/ (CALCUL.) = 187 /MPa/ STRESS (EXPER.) = 189 /MPa/ (CALCUL.) = 189 /MPa/ eps= .7 STRESS (EXPER.) = 186 /MPa/ eps= .8 (CALCUL.) = 188 /MPa/

t [C]	o [min ⁻¹]	STRESSPE	STRESSPC
1000	6	59.7273	59.8895
1000	50	85.456	84.985
1000	500	117.617	118.151
1000	1000	129.256	129.024
900	6	92.507	96.2548
900	50	120.986	128.329
900	500	147.327	166.37
900	1000	164.48	178.114
850	6	107.203	121.149
850	50	139.976	155.578
850	500	169.993	19 4.926
850	1000	188.983	206.872
800	6	142.12	150.634
800	50	164.48	186.434
800	500	192.046	226.309
800	1000	213.487	238.773

TABLE COMPARSION OL EXPERIMENTAL AND CALCULATED RESULTS

PEAK STRAINS

temperature	rate of	strainp	
[K]	twisting [min ⁻¹]	STRAINPE	STRAINPC
1273	6	.197537	.1903
	50	.251786	.257165
	500	.366307	.35636
	1000	.413775	.393024
1173	6	.261415	.266365
	50	.315296	.359915
	500	.461657	.498369
	1000	.549565	.549144
1123	6	.329538	.334584
	50	.466217	.451474
	500	.621253	.624558
	1000	.70535	.687928
1073	6	.443752	.441804
	50	.666054	.594724
	500	.875773	.820567
	1000	.840429	.906523

thus a mathematically correct relationship should be of the form

(16)
$$c = A - B/T$$
 $c = 1,5157 - 998/T$

and eq. (15) transformed to

(17)
$$\sigma = \sigma_p \left[\frac{\varepsilon}{\varepsilon_p} \exp\left(1 - \frac{\varepsilon}{\varepsilon_p} \right) \right]^{A-B/T}$$

Conclusion

Comparison of experimental – STRESSPE – and calculated – STRESSPC – values are given in tables together with STRAINPE and STRAINPC results. The delivered figures 1 to 3 illustrate differences between STRESSPE and STRESSPC according equations (15), (17), respectivelly.

References

- [1] HERTZBERG, R. W., Deformation and Fracture ..., USA, 1983.
- [2] ŽÍDEK, M., KLIBER, J., SCHINDLER, I., Einfluss metallurgischer Faktoren auf den Umformwiderstand mikrolegierte Stähle beim gesteuerten Walzen, Neue Hütte (1990), No. 4, p. 125-129.
- [3] SPITTEL, M., SPITTEL, T., Freiberger Forschungshefte, B-235, 1982.
- [4] ТАМАМОТО, S. et al., Tetsu-to-Hagané, 1981, A49-A52.
- [5] KLIBER, J. at al., Hutnické listy, 40, (1985), č. 2, 90-95.
- [6] SELLARS, C. M. at al., Hot workability, Int. Met. rew., 158, 1-23.
- [7] SCHINDLER, I. at al., Hutnické listy, 45, (1990), č. 4, 261-267.
- [8] KOKADO, J. at al., Steel Research, 56, (1985), No. 12, 619-624.
- [9] ELFMARK, J., Dynamic restoration of 24 % chromium ferritic steels, Czech. J. Phys., B38 (1988), 201-209.
- [10] KLIBER, J. at al., Bestimmung des Grenzumformgrades mit dem Torsionsversuch, Steel Research, 60 (1989), 503-508.
- [11] KLIBER, J. at al., Workability at high conditions, Research work, VŠB Ostrava, 1988.
- [12] CINGARA, A., MCQUEEN, H. J., Int. Symp. of Process., Pensylvania, 1987.